

**AERODYNAMIC LOADING ASSOCIATED WITH SWEPT AND
UNSWEEP SPOILERS ON A FLAT PLATE AT MACH
NUMBERS OF 1.61 AND 2.01**

By DOUGLAS R LORD

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on a 45 sweptback wing at Mach numbers of 1.61 and 2.01 aeroelastic effects on the loading of swept and unswept wings by Douglas R. Lord,
<http://onlinebooks.library.upenn.edu/webbin/book/browse?type=lcsbck&key=Sikorsky%20aircraft&c=x>

A survey has been made of the available literature on the aerodynamic loads associated {Aerodynamic Loads aerodynamic characteristics of a 45 deg swept
<http://citeseerx.ist.psu.edu/viewdoc/summary?doi=10.1.1.226.9075>

Scott Crossfield cracked the Mach 2 threshold flying the Douglas mi and at Mach numbers between 2 hour flight. Immediately after loading the
<http://forum.avsim.net/rss/ccs/1c12-reviews/liveatc.net>

Interactive Aerospace Engineering and Design. Uploaded by Dami n Gh. Info; Research Interests: Aerospace Engineering, Aeronautical Engineering ,
http://www.academia.edu/7067376/Dava_J._Newman_-_Interactive_Aerospace_Engineering_and_Design

A selection has been made of NACA documents that will be useful for light and general aircraft. Also included are some basic airfoil data which will provide historic
http://www.esdu.com/cgi-bin/ps.pl?t=col&p=col_nacaero

The Blohm & Voss P.211.01 was designed to take full advantage of the swept wing's aerodynamic associated with subsonic high load factor the wing
http://en.wikipedia.org/wiki/Sweep_theory

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on Cmo and aerodynamic centre of unswept wing transfer and of the associated boundary layers on a flat plate at Mach numbers are 1.41, 1.51 1.61,
<http://aerade.cranfield.ac.uk/listarccp.php>

Combustion and Aero-Structural Interaction. Associated Faculty. Aerodynamic/dynamic loads and response of wind turbines.
<http://www.aere.iastate.edu/disciplines/aerodynamics-and-aero-structural-interaction/>

NACA Digital Library. Diederichs, H Upton, G B, Mufflers for aeronautic engines, NACA Report 10, 1917, pp. 10. Lucke, Charles Edward Willhofft, Friederich Otto
<http://archive.is/4bEe>

WING AT MACH NUMBERS OF 1.61 AND 2.01 By Douglas R. Lord and spoilers on a flat plate were almost made between a swept and an unswept spoiler located
<http://ntrs.nasa.gov/archive/nasa/casi.ntrs.nasa.gov/19630003034.pdf>

along a vertical heated flat plate.* 22.1,44.2.* ARNOLDI, R.A.* Aerodynamic broadband pressures at Mach numbers between 1.6 and 2.5* NASA
<http://navier.stanford.edu/bradshaw/pbref/r>

Aerodynamic loadings associated with swept and unswept spoilers on a flat-plate at Mach numbers of 1.61 and 2.01. Lord, Douglas R 1956/naca-rm-l55l12
<http://naca.central.cranfield.ac.uk/report.php?NID=6601>

The sectional and total aerodynamic load characteristics and performance degradation of swept wings aerodynamic load associated with the
<http://ntrs.nasa.gov/search.jsp?R=19910036957>

A Case Study of a Glass/Carbon Hybrid Composite Structure. problems associated with the use of innovative effective aerodynamic loads,
http://www.academia.edu/7007882/The_Twist-Bending_Behavior_of_Forward_Swept_Wings_A_Case_Study_of_a_Glass_Carbon_Hybrid_Composite_Structure

Airplane With Unswept Slotted Cruise Wing Airfoil The result is about a cruise speed of Mach=0.78, but with a straight wing. It also means that for the
<http://www.docstoc.com/docs/50313624/Airplane-With-Unswept-Slotted-Cruise-Wing-Airfoil---Patent-6293497>

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investigation made at Mach numbers between 1.36 and 2.01 of the total-pressure A suspended flat plate was used to BenadN.Dleyo fetvns and Douglas R. Lord*
<http://ufdc.ufl.edu/AA00009235/00032>

Aerodynamic Loadings Associated with Swept and Unswept Spoilers on a Flat Plate at Mach Numbers of 1.61 and 2.01: Lord, Douglas R:
<http://ntrs.nasa.gov/search.jsp?R=19930089093>

6. Aerodynamics of 3D Lifting Surfaces through Vortex Lattice Methods 6.1 An Introduction There is a method that is similar to panel methods but very easy t
<http://fr.slideshare.net/franklinnagao/mat-lab-vlm>

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<https://www.scribd.com/doc/98886442/Aerodynamics-for-Engineers-5th-Edition>

Some Aerodynamic Principles for the Design of Swept Wings. Aerodynamics Department, Royal Aircraft Establishment, Farnborough; Available online 18 June 2014. Summary.

<http://www.sciencedirect.com/science/article/pii/B9781483199849500046>

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College of Aviation Technology . 4 years ago. Flag. Understanding Flight. David F. Anderson Scott Eberhardt

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